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Failure Sampling with Optimized Ensemble Approach for the Structural Reliability Analysis of Complex Problems

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- Failure Sampling with Optimized Ensemble Approach for the Structural Reliability Analysis of Complex Problems Christopher Eamon¹, Kapil Patki², and Ahmad Alsendi³
- **Abstract**

 Failure sampling is a structural reliability method based on modified conditional expectation suitable for complex problems for which reliability index-based approaches are inapplicable and simulation is needed. Such problems tend to have non-smooth limit state boundaries or are otherwise highly nonlinear. Previous studies recommended implementation of failure sampling with an extrapolation technique using Johnson's distribution or the generalized lambda distribution. However, what implementation method works best is problem dependent. The uncertainty of which approach provides best results for a particular problem limits the potential effectiveness of the method. In this study, a solution is proposed to this issue that eliminates this uncertainty. The proposed approach is an optimized ensemble that forms a uniquely-weighted solution by utilizing the predictive capability of multiple curves to maximize accuracy for any particular problem. It was found that the proposed approach produces solutions superior to the methods of implementing failure sampling previously presented in the literature.

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Keywords: numerical methods; optimization; uncertainty analysis; reliability analysis

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Introduction

 A large number of structural reliability analysis methods have been proposed in the last several decades. The most common among these can perhaps be grouped into two broad divisions: simulation methods such as Monte Carlo Simulation (MCS) and its many variants; and analytical approaches that compute reliability index as a surrogate for direct evaluation of failure probability. This latter group of methods includes the ubiquitous First Order Reliability Method (FORM), which has become very popular since its introduction in the late 1970s with an adjustment for non-normal random variables (Rackwitz and Fiessler 1978). Although not as frequently used, Second Order Reliability Methods (SORM) have also been proposed (Breitung 1984), as well as similar reliability index based algorithms. Such methods attempt to locate the most probable point of failure (MPP), the peak of the joint probability density function on the failure boundary of the limit state function in standard normal space. Reliability index (*β*) is then typically calculated as the distance from the MPP to the origin, from which a simple transformation to failure probability can be obtained. Although computationally efficient, usually offering vast reductions of computational effort for typical problems over simulation approaches, reliability index based methods cannot guarantee convergence to the true solution, unlike MCS if the sample size is sufficiently increased. Rather, as reliability index methods rely upon an approximation of the limit state boundary at the MPP (a linear approximation in the case of FORM), nonlinearities in standard normal space in which *β* is calculated, either from inherent nonlinearities in the structural response considered for the limit state function, or when non-normal random variables are introduced, will result in some degree of error. Although this error is often small, in some cases, particularly for complex nonlinear limit state functions, it can be unacceptably large (Eamon et al. 2005; Melchers 1999; Chiralaksanakul and Mahadevan 2005; Haldar and Mahadevan 2000). For other complex problems, such as those that are highly nonlinear, discontinuous, or that have multiple local MPPs, search algorithms are sometimes unable to identify the MPP, and the solution process fails completely (Patki and Eamon 2016; Eamon and Charumas 2011).

 Although direct simulation such as MCS may approach the true solution as sample size is increased, the well-known drawback of such methods is the large computational demand for complex engineering problems, particularly such as those requiring finite element analysis for solution. To increase efficiency, numerous variance reduction modifications were proposed to MCS, including stratified sampling (Iman and Conover 1982), importance sampling (Rubinstein 1981; Engelund and Rackwitz 1993), adaptive importance sampling (Wu 1992; Karamchandani et al. 1989), directional simulation (Ditlevesen and Bjerager 1988), dimensional reduction and integration (Acar et al. 2010); subset simulation (Au et al. 2007), and many others. As with any approach, each of these methods has particular disadvantages. For example, stratified sampling, of which perhaps Latin Hypercube (Iman and Conover 1982) is among the most well-known, has not consistently shown significant reductions in computational costs for a variety of problems (Eamon et al. 2005). Importance sampling, as with reliability index methods, which require identification of the MPP, may obtain inaccurate or no solution for complex problems. Although directional simulation is extremely efficient for some problems, particularly for limit state boundaries that are spherical, efficiency is reduced when the limit state boundary takes on a common hyperplanar shape (Engelund and Rackwitz 1993). More recent advancements, however, such as adaptive directional importance sampling, have maintained high efficiency for a variety of non-spherical limit state boundaries when the MPP can be located (Grooteman 2011; Shayanfar 2018). Subset simulation has been developed significantly in the last two decades, producing various alternative implementation approaches. An important consideration with this method, however, is how the importance sampling density is determined, as high variance in the solution can be obtained with sub-optimal selections (Au and Beck 2001; Au et al. 2007).

 Rather than using a more efficient reliability analysis method, the underlying problem itself can be simplified. One systematic way to achieve this is with use of a response surface, where a computationally demanding limit state function evaluation can be replaced with a more simple, analytical surrogate function (Gomes and Awruch 2004; Cheng and Li 2009). The response surface can then be used with a variety of traditional reliability analysis methods to provide fast probabilistic solutions. Alternative surrogate models include those developed from polynomial chaos expansion, Kriging, genetic algorithms, and artificial neural networks (ANN), among others (Gomes 2019; Guo et al. 2020). Adaptive versions of such metamodeling techniques, particularly ANN, have shown to be effective for solving a variety of complex problems (Gomes 2019). However, in some cases, the cost of forming a high- fidelity surrogate model for a complex problem may outweigh the cost of using the original response with a reasonably efficient reliability method to begin with (Eamon and Charumas 2011).

 As an alternative solution for complex reliability problems, Eamon and Charumas (2011) proposed the modified conditional expectation, or failure sampling (FS) method, which was reported to accurately solve various complex limit states with reasonable computational effort. In general, the method uses conditional expectation to sample the complex (generally resistance) portion of the limit state function, then uses a numerical technique to estimate either its probability density function (PDF) or cumulative distribution function (CDF). Failure probability can then computed directly by numerical integration over the failure region. Alternatively, additional resistance data for high reliability problems can be generated by extrapolation, where the original sample is fit to a flexible, multi-parameter curve to extend the tail region. Clearly, the accuracy of this approach is a function of how well the PDF or CDF estimate and resulting curve fit are developed. As might be expected, what implementation method works best is problem dependent. Unfortunately, there is little a priori indication as to what method produces the greatest accuracy for a specific problem. For example, Patki and Eamon (2016) examined various problems with the FS approach and generally recommended data extrapolation using Johnson's Distribution, as they found it produced best results in many cases. However, this is not always true, as using alternative flexible curves such as the generalized lambda distribution or the generalized extreme value distribution produced higher accuracy for some problems. The uncertainty of which approach provides best results for a particular problem limits the potential effectiveness of the method. This issue is the focus of this study. Here, an optimized ensemble approach is developed, with minimal additional computational effort, that forms a uniquely-weighted fit by utilizing the predictive capability of multiple curves that maximizes the accuracy of the FS method for any particular problem.

Summary of Failure Sampling Method

 The FS approach is fully described elsewhere (Eamon and Charumas 2011; Patki and 118 Eamon 2016), whereas a brief summary is provided here. The probability of failure p_f of a limit state function *g* can be calculated by estimating a single-dimensional PDF of *g* and 120 integrating the PDF over the failure region (i.e. where $g < 0$). Direct MCS can be used to generate the sample of *g* used to develop the PDF. Of course, this approach will yield accurate results only when the PDF of *g* can be estimated accurately. However, for typical structural reliability problems, the large majority of the sample generated from MCS is far from the failure region, resulting in a problem for which it is difficult if not impossible to accurately integrate the failure region without a high number of simulations.

126 In the FS approach, the initial limit state function $g(X_i)$, consisting of random variables *Xi*, is reformulated to a new limit state *g**. *g** is expressed in terms of a control random variable *Q*, and the function of remaining random variables (RVs), *R*(*Xj*). Setting *g** to zero to represent the failure boundary, the problem can be written as:

130
$$
g^* = R(X_j) - Q = 0 \tag{1}
$$

 Here *g** is mathematically equivalent to original limit state function *g*. Note that function *R(Xj)* need not be explicitly formed and could be evaluated from FEA or another complex numerical procedure. Moreover, there is no theoretical limitation in the selection of *Q*, although it is often chosen as a load RV in the physical problem for convenience of implementation. For greatest accuracy, it is best that *Q* is selected such that it is statistically independent of the remaining 136 RVs X_i , a stipulation that can be satisfied for at least one RV by nearly all realistic structural reliability problems. Further, it is advantageous to select this variable as that with the highest variance, if possible, which removes its associated uncertainty from the simulated data and may reduce the number of simulations required for the same accuracy. If multiple RVs exist with the same variance, there is no theoretical advantage of selecting one over another as the control 141 variable, and the choice reduces to that of convenience. Once Eq. 1. is formed, values x_i are simulated by a method such as MCS. From Eq. 1., it can be seen that for a particular set of 143 simulated values $R(x_i)$, $q = R(x_i)$. That is, if a value q can be determined to satisfy Eq. 1, that 144 value also equals a datum for the sample of $R(x_i)$. Note for complex problems, this generally requires a non-linear solver to determine *q*, as further discussed in a detailed summary of the 146 procedure given below. A value q is thus determined for each set of simulated values $R(x_i)$, thereby developing an equivalent, single-dimensional data sample for the potentially very 148 complex, multi-variate $R(X_i)$. Once the data sample for $R(X_i)$ is generated, there is no need to evaluate the true response further (e.g. no need for further FEA, if that is how the limit state function is evaluated), and the bulk of the computational effort for a complex problem ends. Next, depending on the solution approach, a PDF or CDF estimate of *R(Xj)* is developed, from 152 which p_f of g^* (and thus of the original limit state function $g(X_i)$) can then be found with a variety of methods. As mentioned above, one approach is numerically integrating over the region with:

155
$$
p_f = \int_{-\infty}^{\infty} F_R(q) f_Q(q) dq
$$
 (2)

156 where F_R refers to the CDF of *R* and f_Q the PDF of *Q*. Due to the sparsity of data with high- reliability problems, numerical integration may lose accuracy. Thus, an alternative approach 158 is to use a flexible curve to represent the data sample for $R(X_i)$, which can be used to extend 159 the tail region indefinitely. Once this is done, p_f can be computed very quickly with any method, such as MCS, for example, as the original, potentially complex function *g** is now represented analytically.

 In summary, the FS approach offers several useful features:1) as the MPP is not used, problems for which the MPP cannot be located, or which is incorrectly located, and thus which are unsolvable or inaccurately solved by reliability index or importance sampling methods, can be addressed; 2) for complex problems of moderate reliability (i.e., reliability index from about 3-5, within the range of typical structural components) that are poorly solved with many other methods, computational effort for FS is relatively low, often on the order of 1000 simulations, for reasonably accurate solutions; 3) the method is simple conceptually as well as to implement. As suggested above, as a simulation-based method, although FS is applicable to any type of problem, it becomes competitive when reliability-index based methods provide no or poor solutions, and for which other simulation methods require an unfeasibly large computational effort. For simpler problem types, reliability index based solutions are generally more efficient. Although the FS approach was previously demonstrated to be effective, as noted above, an existing concern is that it is not clear what method of implementation works best for different problem types. This is the issue that this study attempts to address.

Optimized Ensemble Approach

 Rather than relying upon a single distribution to estimate *p^f* with FS, this study proposes an ensemble approach to maximize the accuracy of the *p^f* calculation from an optimal combination of the multiple available possibilities. Ensembles have used to increase the accuracy of various problems, most notably in metamodeling (Zerpa et al. 2005; Goel et al. 2007; Acar and Rais-Rohani 2009; among others), where in much of this previous work, an ensemble of metamodels is generally represented as a weighted sum of two or more stand- alone metamodels, each separately fitted to the same response using different techniques. The resulting hybrid metamodel takes advantage of the prediction ability of each individual stand-alone metamodel to enhance the accuracy of response predictions.

 In this paper, rather than using traditional metamodels, an ensemble of alternative 188 CDFs of the resistance sample for $R(X_i)$ is established, then the individual CDFs are assigned weight factors depending upon their anticipated accuracy. Using a weighted sum formulation, a unique, problem-specific ensemble of CDFs can thus be formulated as follows:

191
$$
F_{RE} = \sum_{i=1}^{N} w_i F_{RTi}
$$
 (3)

192 where F_{RE} is the final ensemble CDF of *N* stand-alone CDFs F_{RTi} , and w_i is the weight factor of *ith* stand-alone CDF. In this study, three stand-alone CDFs are used, individually fit to the 194 sampled data $R(x_i)$ as discussed in more detail below. The weight factors are subjected to the following constraint:

196
$$
\sum_{i=1}^{N} w_i = 1
$$
 (4)

 The weight factors are determined by a sequential quadratic programming optimization process where the difference between the CDF formed directly from the sampled data, the "true" CDF, and the analytical representation, *FRE*, is minimized. The CDF of the sampled data of *M* total points can be expressed as:

$$
F_R(s) = \frac{s}{1+M} \tag{5}
$$

 where *FR(s)* is the CDF value for datum *s*. Although numerous goodness-of-fit metrics exist, the error between the true CDF and *FRE* is measured in this study using a generalized mean square error (MSE) metric given as:

205
$$
MSE = \frac{1}{M} \sum_{s=1}^{M} (F_{RE}(s) - F_R(s))^2
$$
 (6)

206 The final optimization problem is to find the optimal values of design variables $W = (w_1, w_2,$ *...., wN*) that would

$$
208 \qquad \qquad \text{min} \qquad \text{Err} = \text{MSE} = f(W) \tag{7}
$$

209 s.t. $\sum_{i=1}^{N} w_i = 1$

$$
210 \t\t 0 \leq w_i \leq 1
$$

 Prior to recommending MSE, the authors studied two alternative forms of goodness-of-fit: a log-based criterion to maximize differences in the lower resistance tail, similar to the Anderson- Darling test (Ang and Tang 2007); as well as that based on the linear sum of differences in CDFs, as per Eamon and Charumas (2011). It was found that both could produce better results than MSE in some cases, but significantly worse in other cases, and for problems with no apparently similar characteristics. As opposed to the alternative metrics, which were less reliable overall, it was found that MSE tends to emphasize differences in weights, focusing the weighted response on a single dominant distribution.

 As three stand alone CDFs were used in this study, at the start of the optimization, the 220 initial values for the weight factors *W* are set to 1/3. Once a data sample for $R(X_i)$ is generated, the alternate CDFs are individually fit to the data and the weights are determined. A realized 222 value for $R(X_i)$, r, is then represented as the weighted sum of the values r_i taken from the stand alone CDFs, as determined in accordance with Eq. 3, which results in:

$$
r = \sum_{i=1}^{N} w_i r_i \tag{8}
$$

 As with the original FS method, any reliability method such as MCS or FORM can then be 226 used to quickly determine the p_f estimate of g^* , since $R(X_i)$ is now represented as a fully defined, single dimensional random variable *R*. For example, if MCS were used to solve Eq. 1, a realized value *r* is determined by sampling each of the stand-alone CDFs (using the same random perturbation for each curve per simulation) to produce *N* values, which are then combined in accordance with Eq. 8. Note that, for a complex problem, it is the generation of 231 the data sample for $R(X_i)$ that requires calling the original limit state function (assumed to be a time-consuming FEA code or similar analysis tool), where the complex, multidimensional $R(X_j)$ is transformed to an equivalent single RV via the original FS process. Once the data 234 sample for $R(X_i)$ is established, the computational effort required for all further calculations, including the developments of the alternative CDFs and their weight factors, as well as to estimate *p^f* of *g**, is comparatively trivial (generally seconds of real time on a modern desktop 237 computer). In this study, after the data sample for $R(X_i)$ was established and the ensemble CDF weights determined to complete Eq. 3, MCS was used in conjunction with Eq. 8 to compute *p^f* of *g**. The procedure can be summarized as follows:

The original limit state function is rewritten in the form of Eq. 1.

- 241 1. Values for RVs within $R(X_i)$ are determined by simulation. In this study, direct MCS is used, although other techniques are also possible (Patki and Eamon 2014). 243 A single realized value $R(x_i)$ is thus determined.
- 2. The required value for *Q* necessary to satisfy Eq. 1 is determined. For a simple problem for which the limit state can be explicitly written, *q* can readily be found 246 to be given by $R(x_i)$, as from Eq. 1, $R(x_i) = q$. For more complex problems, where Eq. 1 may be implicit, a nonlinear solver is required to determine *q*. An example

 of this scenario is for problems that involve a finite element procedure, where the relationship between the measured response in the limit state function (for example, stress or displacement) and the value of *q*, as a function of values for the remaining 251 variables x_i , cannot be explicitly established. In practice, Eq. 1 would then be solved by incrementing the value of *q* until the specified failure criterion (such as a stress or displacement limit) is achieved.

- 3. The simulation process is repeated (i.e. steps 2 and 3) until the desired sample size is generated. As with most simulation methods, increasing the sample size generally improves results. However, in this study, 1000 simulations were used. At the conclusion of this step, the actual limit state function (which may be computationally expensive to evaluate) is no longer used in the problem solution.
- 259 4. Since $q = r$ on the failure boundary, the values determined for *Q* also must equal 260 corresponding values for $R(X_i)$; conceptually, values of resistance. Independent 261 curves (CDFs) are then fit to the (1000 point) data sample for $R(X_i)$, essentially reducing the potentially complex resistance function into a representative single random variate (albeit at this point, represented by a set of *i* alternative CDFs). These CDFs are the individual member functions (*FRTi*) of the ensemble.
- 5. Using Eqs. 3-7, weights *wⁱ* are assigned to each CDF by optimization. The final optimized CDF of resistance is then represented by Eq. 3.
- 6. Assuming that *Q* is an RV with known parameters, Eq. 1. can now be explicitly 268 expressed as: $g^* = R - Q$. This simple, two RV limit state function can be solved with any reliability method such as MCS, importance sampling, FORM, SORM, etc, as desired. Here, note that regardless of the method used, *i* different values (i.e. the number of curves used to construct the ensemble) for RV *R* are required for solution. For example, if using a simulation approach such as MCS, for one

 simulation, the same initially generated uniform random value would be used to 274 resolve each of the alternative values r_i from CDFs F_{RTi} . The final realized value r is then given by the weighted sum of the realizations, per Eq. 8.

 To simplify the approach further, it was observed that in many problems a single CDF often dominates the solution with a high weight relative to the other curves considered. In this case, good results can often be obtained by simply using the single dominant curve, forgoing the ensemble. Thus, an effective simplified approach to determine values for *R* can be implemented as follows:

281 $R = R_T$, $w_T \geq Th$

$$
R = Ens, \quad w_T < Th \tag{9}
$$

 where *R^T* is a value of *R* determined from the single dominant distribution; *Ens* is the ensemble approach, given by Eq. 8; *w^T* is the weight of the dominant distribution found from the minimization of MSE per Eqs. 6 and 7; and *Th* is the dominant weight threshold to forgo the ensemble. The choice of *Th* is subjective, representing a desired balance between accuracy and additional complexity, and is best left to the analyst. However, for many problems the authors 288 have found good results for threshold weights as low as $Th = \frac{3}{4}$, as will be discussed below.

 In general, the ensemble approach is intended for problems for which the MPP cannot be located or the failure boundary is not well represented with a FORM/SORM approximation, and direct simulation methods are too costly. Limitations of the ensemble approach include the need for statistical independence of the control variable; the need for a nonlinear solver to 293 set $g^* = 0$ for implicit problems; and most importantly, the need for the surrogate distribution to accurately represent the CDF of *R*. Although seemingly challenging, the latter condition may be satisfied even for relatively complex problems, provided that some prior knowledge of the response is available. For example, if, say, a unique limit state boundary exists such as a truncated or multimodal form, then such a distribution type may be included within the ensemble, using one or more of the numerous representations of such distributions available in the literature. Alternatively, if only one troublesome distribution type is present, it may be taken as the control variable, removing it from the response to be fit.

 Another issue to consider is that, as with any approximate approach, there is no guaranteed method to obtain the error associated with the reliability estimate, which would require knowledge of the true solution beforehand. However, as with most MCS-based approaches, it has been shown that increasing the number of simulations using the FS method produces estimates that tend to converge to the true solution (Eamon and Charumas 2011). Thus, although it is not possible to directly quantify the error associated with the proposed approach, it can at least be ensured that the number of simulations used is sufficient by increasing this number until subsequent results do not significantly differ.

CDFs Considered

311 In this study, CDFs are generated from the $R(x_i)$ data by three methods for use in Eq. 3: use of the generalized lambda distribution (GLD), Johnson's distribution (JSD), and the generalized extreme value distribution (GEV). Although any type and number of CDFs can be included in the analysis, the approach taken here is the use of a smaller number of highly flexible functions.

 The GLD can represent many common distributions such as normal, lognormal, 317 Weibull, and others. It is defined by location (λ_1) , scale (λ_2) , skewness (λ_3) , and kurtosis (λ_4) parameters. Various ways have been developed to estimate these parameters (Karian and Dudewicz 2011; Ozaturk and Dale; Asif and Helmut 2000). The method of moments was used in this study (Karian and Dudewicz 2011). The PDF of the GLD is given by:

321
$$
f_{RT}(x) = \frac{\lambda_2}{[\lambda_3 u^{(\lambda_3 - 1)} + \lambda_4 (1 - u)^{(\lambda_4 - 1)}]} \tag{10}
$$

322 The PDF is expressed in terms of a probability parameter *u*, which is related to random 323 variable x through the inverse of the CDF, the quantile function: $x = Q(u) = \lambda_1 +$ $u^{\lambda_3}-(1-u)^{\lambda_4}$ $\frac{a^{3}-(1-a)^{2}}{\lambda_{2}}$.

 The Johnson's system of distributions is also defined by four parameters and similarly has wide flexibility. This system is composed of multiple normalizing transformations: the bounded, or 'S_B' transformation, which models distributions bounded on either or both ends such as gamma and beta; the semi-bounded 'SL' transformation, which models a lognormal distribution, and the unbounded 'SU' transformation which can model the normal, t, and other distributions. A JSD distribution is defined with two shape parameters *γ* and *δ*, a location 331 parameter ξ , and a scale parameter λ_i . The Johnson's PDF is given by:

332
$$
f_{RT}(x) = \frac{\delta}{\lambda_j \sqrt{2\pi}} g'_r \left(\frac{x - \xi}{\lambda_j}\right) exp\left[-\frac{1}{2}\left(\gamma + \delta g_r \left(\frac{x - \xi}{\lambda_j}\right)\right)^2\right]
$$
(11)

where function $g_r\left(\frac{x-\xi}{\lambda}\right)$ 333 where function $g_r\left(\frac{x-\xi}{\lambda_j}\right)$ is determined by the desired transformation (i.e. either S_B, 334 S_L, or S_U). As with the GLD, multiple methods are available to determine JSD parameters. In 335 this study, the method of quantile estimators was used (Karian and Dudewicz 2011; George 336 2007; Slifker and Shapiro 1980).

 The GEV distribution is described by a location parameter *μ*, a scale parameter *σ,* and a shape parameter *k*. Parameters can be determined using similar methods as those available for fitting the GLD or JSD, such as the method of moments, percentiles, or quantile estimators, the latter of which was used here. The GEV resembles an extreme type distribution and is often used to model the smallest or largest values in a dataset. Its PDF is given as:

342
$$
f_{RT}(x) = \frac{1}{\sigma} \exp\left(-\left(1 + k\frac{(x-\mu)}{\sigma}\right)^{-1/2}\right) \left(\left(1 + k\frac{(x-\mu)}{\sigma}\right)^{-1-\frac{1}{k}}\right)
$$
(12)

 An example result of the ensemble approach implemented with the three curves above is given in Figure 1, which corresponds to example problem 3 discussed below. In the figure, the three individual CDFs are shown, as well as the resulting ensemble. In this problem, the ensemble CDF closely resembles the GEV curve, which dominated the solution (with weights of JSD, 347 GEV, and GLD given by $w_{JSD} = 0.01$, $w_{GEV} = 0.91$, and $w_{GLD} = 0.08$, respectively).

Example Problems

 To illustrate the ensemble approach, several example problems are considered. These include three benchmark reliability problems and two complex engineering problems utilizing nonlinear finite element analysis. Note that the benchmark problems can be expressed algebraically and are thus not of the complexity for which an FS approach is needed, nor for which it would represent the most efficient solution method. However, these are included as theirsolution is readily verifiable and can provide useful information as to the range of problem characteristics for which the ensemble approach can be effective. As the purpose of the example problems is to examine the effectiveness of the ensemble approach rather than produce 358 exact solutions, only 1000 simulations of the actual response function $R(X_i)$ were used to generate the dataset to fit the stand-alone CDFs, even for those problems approaching a reliability index of 4 (with corresponding failure probability of about 1:30,000). Although this produced reasonably accurate solutions for the problems considered, additional accuracy can generally be obtained with additional simulations. In this study, once the ensemble CDF was 363 formed, MCS was used to quickly compute p_f of the resulting simple two RV limit state g^* = $B - Q$ (using $1x10⁶$ -10⁷ simulations, depending on reliability level), although a less expensive alternative method such as FORM would have produced solutions with no significant loss of 366 accuracy as well. Results are reported in terms of reliability index β with the standard normal 367 transformation $\beta = -\Phi^{-1}(p_f)$.

Problem Set 1: Benchmark Limit State Functions

 Engelund and Rackwitz (1993) proposed a series of unique limit state functions for method evaluation. Two of these cases, a multiple reliability index function and a maximum function, were found by the authors of this study to be among the more difficult to solve accurately with traditional analytical methods such as FORM, and are evaluated below. In each of the benchmark problems, the reference solution (taken as the "exact" solution) was 375 computed using a sample size of $1x10⁹$ with MCS.

 The multiple reliability index case represents a hyperbolic function with two reliability indices, and is given as:

$$
378 \t\t g = x_1 x_2 - k \t\t(13)
$$

 where *x¹* (taken as control variable) and *x²* are normal RVs having mean values of 78064.4 and 0.0104, with corresponding standard deviations of 11709.7 and 0.00156, respectively, and constant *k* was taken as 480, 240, and 160 in this study to produce different reliability levels to investigate. As shown in Table 1, although accuracy was effected when limiting to 1000 simulations, the ensemble produced superior results to the GEV and GLD distributions alone, 384 while the JSD fit produced no failures (weights: $w_{GEV} = 0.99$; w_{JSD} and $w_{GLD} < 0.01$ for all values of *k*). Note that even if a CDF was found to be ineffective by itself, such as the JSD and GLD in this case, it was still included in the ensemble. However, it was found that curves that produced no failures resulted in an insignificant weight (i.e. near zero) in the ensemble, indicating, as expected, a poor fit to the data. Traditional FORM and MCS solutions are 389 provided for comparison. As expected, MCS could produce no failures (i.e. $p_f = 0$) when limiting to 1000 simulations, while two different FORM algorithms were used to solve the problem (based on those of Rackwitz and Fiessler (1978) and Ayyub and Haldar (1984)), which resulted in different MPPs and correspondingly different reliability indices. Here it should be pointed out that the purpose of providing the FORM and MCS comparison solutions is not to suggest that all of the many available variants of these approaches are unable provide a satisfactory solution, but rather to illustrate that the example problems are reasonably challenging and provide some difficulty for traditional approaches.

 The maximum function, essentially a parallel system, is expressed as the maximum of several sub-functions, and results in a non-smooth limit state boundary. It is given as:

$$
g = max(g_1, g_2, g_3, g_4) \tag{14}
$$

where:

401 $g_1 = 2.677 - u_1 - u_2$

$$
g_2 = 2.500 - u_2 - u_3
$$

$$
g_3 = 2.323 - u_3 - u_4
$$

404 $q_4 = 2.250 - u_4 - u_5$

 All *uⁱ* are standard normal random variables (*u¹* taken as control variable). The reference 407 solution was obtained from a sample size of $1x10⁹$ using MCS. As shown in Table 2, FORM could not converge to a solution, while MCS (1000 simulations) produced no failures and GLD and GEV could not successfully fit the resistance data. The ensemble thus defaulted to the JSD 410 approach, which produced a reasonably low error (weights: $w_{JSD} = 0.98$; w_{GEV} and $w_{GLD} \approx$ 0.01).

 A third analytical problem, a circular limit state function, is presented that considers non-normal random variables. In this example, random variables are considered to be either both lognormal or both extreme type I. The limit state function is given as:

415 $g = r^2 - x_1^2 - x_2^2$ (15)

416 where r^2 is taken as 7.0 for the lognormal case and 9.0 for the extreme I case. For both cases, the means and standard deviations of the random variables were taken as 1.0 and 0.25, respectively (*x¹* taken as control variable). Results are given in Table 3, where it can be seen that the ensemble produced better results than the three individual distributions considered 420 (weights: $w_{GEV} = 0.99$; w_{GLD} and $w_{JSD} < 0.01$ for both types of RVs).

Problem 2: Nonlinear Truss with Complex Random Variable Set

 This problem is based on that described in Eamon and Charumas (2011), and is meant to represent complexity within the range of that for which the FS approach was intended. As shown in Figure 2, a 10 member truss with a non-linear material model is subjected to a load *P*. Solution of the problem cannot be achieved with a closed-form analytical expression, and a finite element code (ABAQUS Version 6.11-2) using 10 two-node truss elements (8 total non- zero degrees of freedom) was used to evaluate the response, as solved using the (implicit) Newton-Raphson approach with a residual convergence criteria of 0.005. The material assumed was steel, with a bilinear stress-strain curve and an elastic modulus *E* of 200 GPa. Random 431 variables are the cross-sectional area (*A*), yield stress (σ ^{*y*}), and post-yield modulus (*E*₂) of each truss member, and load (*P*) (taken as control variable). Random variables are taken to have different types of distribution, level of variance, and correlation, as summarized in Table 4. Note that for the normal RVs *A* and *E2*, negative values are theoretically possible during the simulation, potentially producing a physically impossible problem as well as a failed FEA 436 solution attempt. Since only 1000 simulations were used to generate the data sample for $R(X_i)$, this did not occur (and represents an improbable result, as negative values occur at 10 and 20 standard deviations from the means of *A* and *E2*, respectively). However, for cases in which this would be a concern, alternative distributions or appropriate truncated RV types could be used.

 The failure criterion was defined as the state where the stress in member 1 reaches its yield stress. The resulting limit state function is given as:

443
$$
g = \sigma_{y1} - \sigma_1(P, \sigma_{yj}, E_{2i}, A_i)
$$
 for $i = 1$ to $10, j = 2$ to 10 (16)

The reference solution (β =3.50) was obtained from 1 x 10⁶ MCS simulations. Note that although the limit state function was evaluated with a sample size of 1000, (the "nominal" number of calls), the actual number of function calls using the ensemble approach exceeded 447 this value, due to the iterative process needed to find the root of $g^* = 0$, as shown in Table 5; such iteration is not required for the explicitly formulated response functions in problem set 1, for which roots can be determined analytically. Here a version of the bracketing method was used for solution (Suhadolnik 2012), with an error tolerance of 2%. For comparison, and to verify the suitability of problem complexity, a FORM solution was also attempted, and failed to provide a solution, as the MPP could not be located, even after using several different search algorithms and different starting points. Similarly, as expected, no solution could be obtained from MCS when limiting the actual number of function calls to that of the ensemble approach. 455 In this problem, the GEV dominated the solution (weights: $w_{GEV} = 0.753$; $w_{GLD} = 0.0011$; w_{JSD} $456 = 0.246$, and thus the simplified threshold method was used for illustration in lieu of the complete ensemble, from which good results were obtained.

Problem 3: Highly Nonlinear Column with Large Random Variable Set

 This problem is based on that given by Alsendi and Eamon (2020). It represents a reinforced concrete bridge pier column subjected to a blast load initiated at the column base. The column base is fixed and the top is constrained by a beam element representing the pier beam cap, which is connected to two additional columns forming the pier structure, which are also modeled with beam elements (not shown in Figure 3 for clarity). The column is 3 m high and 760 mm square, and reinforced with 24 #8 vertical bars (6 bars per face) and #4 ties spaced at 300 mm. An axial load is applied to the column representing the dead load portion of a two- lane, two-span (15 m per span) continuous bridge with a superstructure of five steel girders (spaced at 2.7 m) and a 240 mm thick reinforced concrete deck that the column supports.

 Resistance random variables are concrete compressive strength (*f'c*), yield stress (*Fyl*; F_{vt} , Young's modulus (E_l ; E_t), and tangent modulus (E_{Tl} ; E_{Tt}) of the longitudinal bars (*l*) and ties (*t)*. Random variables associated with each longitudinal bar are taken as independent of each other, while those for transverse bars are taken as perfectly correlated (for random variables of the same type), resulting in 75 total random variables characterizing steel uncertainties, as summarized in Table 6. Load random variables are those of the bridge gravity load and blast load. Gravity (dead) load random variables are those of the prefabricated items such as the steel girders and diaphragms (*DLp*); the cast-in place items such as the deck and barriers (*DLs*); and the wearing surface (*DLw*). Statistical parameters for concrete and steel yield strength are taken from Nowak and Szerszen (2003), while statistics for steel stiffness are taken from Galambos and Ravindra (1978), and those for gravity loads are taken from Nowak (1999). The blast load random variables are the equivalent mass of TNT (kg) (*Qw*) and the net 481 equivalency factor (Q_e) (taken as control variable), where variation in Q_w is meant to account uncertainty in charge weight construction and *Q^e* accounts for uncertainty in the resulting blast pressure. Statistical parameters for these two random variables are taken from Netherton and Stewart (2010). All distributions are reported to be normal except *Qe*, which is triangular. In total, 81 random variables were considered.

 In this problem, failure is defined as a horizontal displacement of the column base that exceeds 4.3 mm within the first 8 ms after the blast initiates (a rate of deformation associated with subsequent column collapse). The resulting limit state function is given as:

489 $g = 4.3 - D(R, Q)$ (24)

 where *D* is the maximum displacement of the column base resulting from the blast at a time of 8 ms, and *R* and *Q* are the sets of resistance and load random variables given in Table 6. Response *D* was evaluated using a large strain, large displacement FEA approach, where the model had approximately 4800 8-node hexahedral (for concrete) and 2-node beam (for reinforcement) elements. Concrete was represented with the Holmquist-Cook model (Holmquist et al. 1993), which accounts for crushing and cracking due to accumulated damage under high rates of strain. Reinforcing steel is modeled with a kinematic, bi-linear material model, while the blast load time-pressure history was represented by the CONWEP method (Hyde 1988). The problem was solved on 4 CPUs in parallel using the finite element code LS- DYNA. As this problem is significantly nonlinear, with the displacement response *D* fairly sensitive to parameter changes, as shown in Table 7, about twice the number of iterations were required to determine the root of the limit state function than for the nonlinear truss problem previously studied. Consideration of a more sophisticated root finding algorithm than the bracketing method used may further reduce this requirement, however. As shown in Table 7, 504 the ensemble provided best results overall (weights: $w_{GEV} = 0.91$; $w_{GLD} = 0.08$; $w_{JSD} = 0.01$), whereas GLD could not be fit to the resistance data, JSD produced a relatively high error, and FORM could not converge to a solution.

Conclusion and Recommendations

 Previous formulations of the failure sampling method were limited by uncertainty with the method of implementation, where the approach with greatest accuracy is highly problem- specific. In this study, a solution was proposed to this issue that reduces this uncertainty and increases the effectiveness of the method regardless of the problem considered. It was found that the ensemble approach is suitable for complex responses and highly nonlinear limit state boundaries. It was further found that the approach is expected to produce solutions at least as good, and often better, than the best single failure sampling implementation method previously presented in the literature. Although the ensemble method is thus recommended for implementation of the failure sampling approach, significant opportunities exist for further development. Among these, more rigorously exploring alternative goodness-of-fit metrics,

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649 Table 1. Hyperbolic Function Results.

650 *NF = no failures. **Results given for alternate algorithms; see text. For FORM, no. of call depends on

651 problem and algorithm; range is given. ***Increasing the number of simulations to 2000 produced β = 4.96 (0.4% error). $(0.4%$ error).

653

654 Table 2. Maximum Function Results.

655 *Solution could not converge. **Could not fit the resistance data.

656

657

658 Table 3. Circular Limit State Results.

669

670 Table 4. Random Variables for Truss Problem.

| Taon – Kangolii vallaonos tol Trass Froorem. | | | | | | | | |
|--|-------|---------------------|---------|-------|--------------|--|--|--|
| Random Variable | Total | Mean | COV^* | ∩** | Distribution | | | |
| area (A) | | 1290 mm^2 | 0.05 | 0.30 | normal | | | |
| yield stress (σ_v) | 10 | 345 MPa | 0.15 | 0.50 | lognormal | | | |
| post-yield modulus (E_2) | 10 | 8280 MPa | 0.10 | 0.70 | normal | | | |
| load (P) | | 85 kN | 0.35 | $- -$ | extreme I | | | |

671 *Coefficient of variation.
672 **Correlation coefficient

**Correlation coefficient between random variables of the same type between different truss members.

675

676

677 Table 5. Truss Problem Results.

678 679

680 Table 6. Random Variables for Column Problem.

| Random Variable (RV) | Total | Nominal value | Bias factor* | $COV**$ |
|--|-------|-----------------|-----------------------|---------|
| Resistance RVs | | | | |
| Concrete strength (fc) | | 41 MPa | 1.15 | 0.15 |
| Yield stress, long. bars (F_{vl}) | 24 | 414 MPa | 1.14 | 0.05 |
| Yield stress, ties (F_{vt}) | | 276 MPa | 1.145 | 0.05 |
| Young's Modulus, long. bars (E_l) | 24 | 200 GPa | 1.0 | 0.04 |
| Young's Modulus, ties (E_t) | 1 | 200 GPa | 1.0 | 0.04 |
| Tangent modulus, long. bars $(E_{\tau l})$ | 24 | 20 GPa | 1.0 | 0.04 |
| Tangent modulus, ties (E_{Tt}) | | 20 GPa | 1.0 | 0.04 |
| | | | | |
| Load RVs | | | | |
| Weight, prefab items (DL_p) | | 67 kN | 1.03 | 0.08 |
| Weight, cast in place items (DLs) | | 387 kN | 1.05 | 0.10 |
| Weight, wearing surface (DL_w) | | 134 kN | Mean= 89 mm | 0.25 |
| Charge weight (Q_w) | | 47 kg | 1.000 | 0.10 |
| Equivalency factor (Q_e) | | 1.00 | $Mode=0.82$ | 0.36 |

*Ratio of mean to nominal value. **Coefficient of variation. Not available for DL_w and Q_e as shown.

682

- 683
- 684 Table 7. Column Problem Results.
- 685

Figure 1. Example Ensemble of CDFs .

Figure 3. Column Subjected to Blast Load.